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SUBROUTINE UEL(RHS,AMATRX,SVARS,ENERGY,NDOFEL,NRHS,NSVARS,
1 PROPS,NPROPS,COORDS,MCRD,NNODE,U,DU,V,A,JTYPE,TIME,
2 DTIME,KSTEP,KINC,JELEM,PARAMS,NDLOAD,JDLTYP,ADLMAG,
3 PREDEF,NPREFD,LFLAGS,MLVARX,DDLMAG,MDLOAD,PNEWDT,
4 JPROPS,NJPROP,PERIOD)

C
INCLUDE 'ABA_PARAM.INC'
PARAMETER ( ZERO=0.D0, HALF=0.5D0, ONE=1.D0, TWO=2.D0, THREE=3.D0,
*             FOUR=4.D0, FIVE=5.D0, SIX=6.D0, PI=3.14159D0)

C
DIMENSION RHS(MLVARX,*),AMATRX(NDOFEL,NDOFEL),
1 SVARS(NSVARS),ENERGY(8),PROPS(*),COORDS(MCRD,NNODE),
2 U(NDOFEL),DU(MLVARX,*),V(NDOFEL),A(NDOFEL),TIME(2),
3 PARAMS(3),JDLTYP(MDLOAD,*),ADLMAG(MDLOAD,*),
4 DDLMAG(MDLOAD,*),PREDEF(2,NPREFD,NNODE),LFLAGS(*),
5 JPROPS(*)
DIMENSION SRESID(12)
REAL Kg(12,12), Fg(12)
REAL Tlb(6,12),Tgl(12,12),TlbT(12,6),TglT(12,12)
REAL AM, r, Tr, h, hl, L, Ar, AL, I, S, rg, Ecp, E, Kv0,
*      Fc, uc, Er, As, Is, Pe, Fcr, ucr, ke, k0, Kr, Kt
REAL Fcrn, Fmax, umax, fcn, ucn
INTEGER n

C
C   UEL SUBROUTINE FOR A ELASTOMERIC BEARING ELEMENT
C

C   SRESID - stores the static residual at time t+dt
C   SVARS - In 1-6, contains the static residual at time t
C           upon entering the routine. SRESID is copied to
C           SVARS(1-6) after the dynamic residual has been
C           calculated.
C           - For half-increment residual calculations: In 7-12,
C             contains the static residual at the beginning
C             of the previous increment. SVARS(1-6) are copied
C             into SVARS(7-12) after the dynamic residual has
C             been calculated.

C **** Parameters ****
C ****

qRubber=PROPS(1)          !Yield strength of rubber
uy=PROPS(2)                !Horizontal yield displacement
G=PROPS(3)                 !Shear modulus of rubber
Kbulk=PROPS(4)              !Bulk modulus of rubber
D1=PROPS(5)                !Internal diameter
D2=PROPS(6)                !External diameter
t=PROPS(7)                 !Thickness of single shim plate
ts=PROPS(8)                !Thickness of single rubber layer
kc=PROPS(9)                !Cavitation parameter
phi=PROPS(10)               !Maximum allowable damage in the rubber
ac=PROPS(11)                !Strength degradation parameter
sDratio=PROPS(12)            !Shear distance
m=PROPS(13)                !Mass of the bearing
cd=PROPS(14)                !viscous damping parameter
tc=PROPS(15)                !Cover thickness

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n=JPROPS(1)                                !Total number of rubber layers
C
C ****
C          Derived parameters   k
C ****
C      Geometric and material parameters
AM=m/TWO                                     !Mass at one node
rd=D2/D1                                       !Diameter ratio
Tr=n*t                                         !Total rubber thickness
h=n*t+(n-1)*ts                               !Height of rubber and steel shims
L=n*t+(n-1)*ts                               !Total height of bearing
Ar=pi*((D2+0.5*tc)**2-D1**2)/4             !Initial bonded area
I=(pi/64)*((D2+0.5*tc)**4-D1**4)           !Moment of interia
S=(D2-D1)/(4.0*t)                            !Shape factor
F=(rd*rd+1.0)/((rd-1.0)*(rd-1.0)) + (1.0+rd)/((1.0-rd)*LOG(rd)) !Dia mod factor
rg=SQRT(I/Ar)                                 !Radius of gyration
C      Elastic stiffness
Ecp=1.0/((1.0/(6.0*G*S*S*F))+(4.0/(3.0*Kbulk))) !Compression modulus
E=3*G                                         !Elastic modulus
Kv0 = (Ar*Ecp/Tr)                            !Pre-cavitation stiffness
C      Cavitation parameters
Fc=3*G*Ar                                     !Initial yield stress
uc=Fc/Kv0                                      !intial yield strain
C      Buckling parameters
Er=Ecp/3                                       !Rotation modulus
As=Ar*h/Tr                                     !Adjusted shear area of bearing
Is=I*h/Tr                                      !Adjusted moment of interia
Pe=pi*pi*Er*Is/(h**2)                         !Euler buckling load of bearing
Fcr=SQRT(Pe*G*As)                            !Critical buckling load in compression
ucr=Fcr/Kv0                                     !Critical buckling deformation
C      Horizontal motion
qYield=qRubber                                !Yield strength of bearing(approximation)
ke=G*Ar/Tr                                     !Elastic modulus of rubber
k0=qRubber/uy                                  !Initial stiffness of hysteretic(approx)
C      Rotation
Kr=Er*Is/Tr
C      Torsion
Kt=G*(2*Is)/Tr
C
C      PRINT*, 'G', G, 'Kbulk', Kbulk, 'Ar', Ar, 'Tr', Tr,
*        'h', h, 'S', S, 'F', F
PRINT*, 'Horizontal Motion:'
PRINT*, 'k0', k0, 'ke', ke, 'qYield', qYield
PRINT*, 'Vertical Motion:'
PRINT*, 'Ecp', Ecp, 'E', E, 'rg', rg, 'Kpre', Kv0,
*        'Fc', Fc, 'uc', uc, 'Fcr', Fcr, 'ucr', ucr
C
C ****
C          Initialize AMATRX, RHS, SRESID
C ****
C
DO K1 = 1, NDOFEL
  SRESID(K1) = ZEROF
  DO KRHS = 1, NRHS

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RHS(K1,KRHS) = ZERO
END DO
DO K2 = 1, NDOFEL
    AMATRIX(K2,K1) = ZERO
END DO
END DO

C
C
DO K1=1,12
    DO K2=1,12
        Kg(K1,K2)=ZERO
    ENDDO
ENDDO

C
DO K1=1,12
    Fg(K1)=ZERO
ENDDO

C ****
C           Analysis cases not considered here
C ****
IF (NDLOAD.NE.0) THEN
    WRITE(7,*)
        'Error:Element loads not implemented'
    CALL XIT
END IF

C
IF (NRHS.EQ.2) THEN
    WRITE(7,*)
        'Error:Riks solution not supported by element'
    CALL XIT
END IF

C ****
C           Analysis Cases
C ****
C
IF (LFLAGS(3).EQ.1) THEN
    Call Transformation
    Call ForceStiffness
C     Normal incrementation
    IF (LFLAGS(1).EQ.1 .OR. LFLAGS(1).EQ.2) THEN
C         *STATIC
        AMATRIX(1:12,1:12) = Kg
        IF (LFLAGS(4).NE.0) THEN
            SRESID=MATMUL(Kg,DU(1:12,1))
            RHS(1:12,1)=RHS(1:12,1)-SRESID
        ELSE
            SRESID=Fg
            RHS(1:12,1)=RHS(1:12,1)-SRESID
        ENDIF
    ELSEIF (LFLAGS(1).EQ.11 .OR. LFLAGS(1).EQ.12) THEN
C         *DYNAMIC
        ALPHA=PARAMS(1)
        BETA=PARAMS(2)
        GAMMA=PARAMS(3)

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C
    DADU = ONE/(BETA*DTIME**2)
    DVDU = GAMMA/(BETA*DTIME)

C
    DO K1=1,NDOFEL
        AMATRX(K1,K1)=AM*DADU
        RHS(K1,1) = RHS(K1,1)-AM*A(K1)
    ENDDO
    AMATRX(1:12,1:12)=AMATRX(1:12,1:12)+(ONE+ALPHA)*Kg
    SRESID=Fg
    RHS(1:12,1)=RHS(1:12,1)
    *      -((ONE+ALPHA)*SRESID-ALPHA*SVARS(1:12))
    DO K1=1, NDOFEL
        SVARS(K1+12)=SVARS(K1)
        SVARS(K1)     =SRESID(K1)
    ENDDO
    ENDIF
ELSEIF (LFLAGS(3).EQ.2) THEN
    Stiffness Matrix
    AMATRX(1:12,1:12) = Kg
ELSEIF (LFLAGS(3).EQ.3) THEN !For damping case
    Not required if material of bearing do not provide damping
ELSEIF (LFLAGS(3).EQ.4) THEN
    Mass Matrix
    DO K1=1, NDOFEL
        AMATRX(K1,K1)= AM
    ENDDO
ELSEIF (LFLAGS(3).EQ.5) THEN
    Half-increment residual calculation
    ALPHA = PARAMS(1)
    SRESID=Fg
    RHS(1:12,1)=RHS(1:12,1)-AM*A-(ONE+ALPHA)*SRESID
    *          +HALF*ALPHA*(SVARS(1:12)+SVARS(13:24))
ELSEIF (LFLAGS(3).EQ.6) THEN
    Initial acceleration calculation
    DO K1=1, NDOFEL
        AMATRX(K1,K1)= AM
    ENDDO
    SRESID=Fg
    RHS(1:12,1)=RHS(1:12,1)-SRESID
    DO K1=1,NDOFEL
        SVARS(K1)=SRESID(K1)
    ENDDO
ELSEIF (LFLAGS(3).EQ.100) THEN
    Output for perturbations
    AMATRX(1:12,1:12) = Kg
    IF (LFLAGS(1).EQ.1 .OR. LFLAGS(1).EQ.2) THEN
        *STATIC
        SRESID=MATMUL(Kg,DU(1:12,1))
        RHS(1:12,1)=RHS(1:12,1)-SRESID
        DO KVAR = 1, NSVARS
            SVARS(KVAR) = ZERO
        END DO
        SVARS(1:12) = RHS(1:12,1)
    ELSEIF (LFLAGS(1).EQ.41) THEN

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C *FREQUENCY
DO KRHS=1,NHRS
    SRESID=MATMUL(Kg,DU(:,KRHS))
    RHS(1:12,KRHS)=RHS(1:12,KRHS)-SRESID
ENDDO
DO KVAR=1,NSVARS
    SVARS(KVAR)=ZERO
ENDDO
SVARS(1:12)=RHS(1:12,1)
ENDIF
ENDIF

C ****
C          SUBROUTINES
C ****
CONTAINS
C-----+
C      Subroutine for transformation matrices
C-----+
SUBROUTINE Transformation
REAL r, TEMP(3,3)
r=SQRT((COORDS(1,2)-COORDS(1,1))**2+(COORDS(2,2)-
*           COORDS(2,1))**2+(COORDS(3,2)-COORDS(3,1))**2)
C Local to basic transformation matrix: Tlb
DO K1=1, 6
    DO K2=1,12
        Tlb(K1,K2)=ZERO
    ENDDO
ENDDO
C Populate the Tlb matrix
DO K1=1, 6
    Tlb(K1,K1)=-1.0
    Tlb(K1,K1+6)=1.0
ENDDO
Tlb(2,6) = -sDratio*L;
Tlb(2,12) = -(1.0 - sDratio)*L;
Tlb(3,5) = sDratio*L;
Tlb(3,11) = (1.0 - sDratio)*L;
C Transpose of Tlb
TlbT=TRANSPOSE(Tlb)

C Local to basic transformation matrix: Tgl
DO K1=1, 12
    DO K2=1,12
        Tgl(K1,K2)=ZERO
    ENDDO
ENDDO
C
IF (ABS(COORDS(1,2)-COORDS(1,1)) .GE. 0.99*r) THEN
    TEMP=RESHAPE((/1,0,0,0,1,0,0,0,1/),SHAPE(TEMP))
ELSEIF (ABS(COORDS(2,2)-COORDS(2,1)) .GE. 0.99*r) THEN
    TEMP=RESHAPE((/0,-1,0,1,0,0,0,0,1/),SHAPE(TEMP))
ELSEIF (ABS(COORDS(3,2)-COORDS(3,1)) .GE. 0.99*r) THEN
    TEMP=RESHAPE((/0,-1,0,0,0,-1,1,0,0/),SHAPE(TEMP))
ELSE

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        WRITE(7,*)
*           'Inclined orientation of
*              bearing is not supported'
        CALL XIT
    ENDIF
C   Populate the Tgl Matrix
    DO K1=1,4
        Tgl(3*(K1-1)+1:3*(K1-1)+3,3*(K1-1)+1:3*(K1-1)+3)=TEMP
    ENDDO
C   Transpose of Tgl
    TglT=TRANSPOSE(Tgl)
END SUBROUTINE

C-----  

C       Subroutine for forces and stiffness matrices  

C-----  

SUBROUTINE ForceStiffness
    REAL ub(6), vb(6), dub(6), fb(6), Kb(6,6)
    REAL uh, vh, Delta, Ard, ratio
    REAL z(2), zC(2), dzdu(2,2), fbc(2), delta_z(2), Df(2,2)
    REAL DBL, tol
    REAL beta, gamma, zNrm, zCNrm, delta_zNrm, tmp1, tmp2, tmp3
    INTEGER iter, maxIter
    DO K1=1,6
        fb(K1)=ZERO
        ub(K1)=ZERO
        vb(K1)=ZERO
        dub(K1)=ZERO
        DO K2=1,6
            Kb(K1,K2)=ZERO
        ENDDO
    ENDDO
    DO K1=1,2
        z(K1)=ZERO
        zC(K1)=ZERO
        delta_z(K1)=ZERO
        DO K2=1,2
            dzdu(K1,K2)=ZERO
            Df(K1,K2)=ZERO
        ENDDO
    ENDDO
    ub=MATMUL(Tlb,MATMUL(Tgl,U))
    vb=MATMUL(Tlb,MATMUL(Tgl,V))
    dub=MATMUL(Tlb,MATMUL(Tgl,DU(1:12,1)))
    zC=SVARS(26:27)
    z=SVARS(26:27)
    ke=SVARS(28)
    uh=SQRT(ub(2)**2+ub(3)**2)
    vh=SQRT(vb(2)**2+vb(3)**2)
    zCNrm=SQRT(zC(1)**2+zC(2)**2)
C   Update parameters
    Kv=Kv0*(1.0/(1.0+(3.0/pi**2)*((uh/rg)**2)))
    IF (uh .GT. ZERO) THEN
        uc=Fc/Kv
    ENDIF
    umax = MAX(uc, SVARS(25))
    Fmax=Fc*(1.0+(1.0/(Tr*kc))*(1-EXP(-kc*(umax-uc))))

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Fcn=Fc*(1-phi*(1-EXP(-ac*(umax-uc)/uc)))
ucn=Fcn/Kv
Delta = 2*acos(uh/D2);
Ard=(D2*D2/4)*(Delta-sin(Delta))
ratio=Ard/Ar
IF (ratio .GT. 0.2) THEN
  Fcrn=Fcr*Ard/Ar
ELSE
  Fcrn=0.2*Fcr
ENDIF
ucr = Fcrn/Kv

C
C
C      axial force and stiffness in basic x direction
IF (ub(1) .LE. ucr) THEN
  Kb(1,1)=Kv/1000
  fb(1)=Fcrn+Kb(1,1)*(ub(1)-ucr)
ELSEIF (ub(1) .LE. ucn) THEN
  Kb(1,1)=Kv
  fb(1)=Kb(1,1)*ub(1)
ELSEIF (ub(1) .LE. umax) THEN
  Kb(1,1)=(Fmax-fcn)/(umax-ucn)
  fb(1)=Fcn+Kb(1,1)*(ub(1)-ucn)
ELSE
  Kb(1,1)=(Fc/Tr)*EXP(-kc*(ub(1)-uc))
  fb(1)=Fc*(1.0+(1.0/(Tr*kc))*(1-EXP(-kc*(ub(1)-uc))))
  umax=ub(1)
ENDIF
Shear forces and stiffnesses in basic y and z direction
DBL=1.0E-32
IF (SQRT(dub(2)**2+dub(3)**2) .GT. ZERO) THEN
Calculate hysteretic evolution parameter z using Newton-Raphson
iter=0
maxIter=100
tol=1E-7
beta=0.1
gamma=0.9

C
10
CONTINUE
zNrm=SQRT(z(1)**2+z(2)**2)
IF (zNrm .LE. DBL) THEN
  zNrm=DBL
ENDIF
tmp1=SIGN(gamma, z(1)*dub(2))+beta
tmp2=SIGN(gamma, z(2)*dub(3))+beta
tmp3=z(1)*dub(2)*tmp1+z(2)*dub(3)*tmp2
function and derivative
fbc(1)= z(1)-zc(1)-(1.0/uy)*(dub(2)-z(1)*tmp3)
fbc(2)= z(2)-zc(2)-(1.0/uy)*(dub(3)-z(2)*tmp3)

C
Df(1,1)=1.0+(1.0/uy)*(2*z(1)*dub(2)*tmp1+z(2)*dub(3)*tmp2)
Df(2,1)=(tmp1/uy)*z(2)*dub(2)
Df(1,2)=(tmp2/uy)*z(1)*dub(3)
Df(2,2)=1.0+(1.0/uy)*(z(1)*dub(2)*tmp1+2*z(2)*dub(3)*tmp2)

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C           Issue wanring if diagonal of derivative Df is zero
IF ((ABS(Df(1,1)) .LE. DBL) .OR. (ABS(Df(2,2)) .LE. DBL)) THEN
  WRITE(7,*)
*           'WARNING: LeadRubberX -zero Jacobian in Newton-
*           Raphson scheme for hysteretic evolution parameter z'
  CALL XIT
END IF

C           advance one step
delta_z(1)=(fbc(1)*Df(2,2)-fbc(2)*Df(1,2))
*           /(Df(1,1)*Df(2,2)-Df(1,2)*Df(2,1))
delta_z(2) = (fbc(1)*Df(2,1)-fbc(2)*Df(1,1))
*           /(Df(1,2)*Df(2,1)-Df(1,1)*Df(2,2))
z = z-delta_z
iter = iter+1
delta_zNrm=SQRT(delta_z(1)**2+delta_z(2)**2)
IF ((delta_zNrm .GE. tol) .AND. (iter .LT. maxIter)) GOTO 10

C           Issue wanring if Newton-Raphson scheme did not converge
IF (iter .GE. maxIter) THEN
  WRITE(7,*)
*           'WARNING: LeadRubberX -The Newton-Raphson Scheme
*           for z did not converge after ', iter, ' iterations
*           and', delta_zNrm, 'Norm'
  CALL XIT
END IF

C           // get derivative of hysteretic evolution parameter
delta_z = z-zC;
IF (ABS(dub(2)) .GT. ZERO) THEN
dzdu(1,1) = delta_z(1)/dub(2)
dzdu(2,1) = delta_z(2)/dub(2)
ENDIF
IF (ABS(dub(3)) .GT. ZERO) THEN
dzdu(1,2) = delta_z(1)/dub(3)
dzdu(2,2) = delta_z(2)/dub(3)
ENDIF
set shear force
fb(2)=cd*vb(2)+qYield*z(1)+ke*ub(2)
fb(3)=cd*vb(3)+qYield*z(2)+ke*ub(3)
set tangent stiffness
Kb(2,2) = cd/DTIME+qYield*dzdu(1,1) + ke
Kb(2,3) = qYield*dzdu(1,2)
Kb(3,2) = qYield*dzdu(2,1)
Kb(3,3) = cd/DTIME+qYield*dzdu(2,2) + ke

C           ELSE
fb(2)=cd*vb(2)+qYield*z(1)+ke*ub(2)
fb(3)=cd*vb(3)+qYield*z(2)+ke*ub(3)
set tangent stiffness
Kb(2,2) = cd/DTIME+qYield*dzdu(1,1) + ke
Kb(2,3) = qYield*dzdu(1,2)
Kb(3,2) = qYield*dzdu(2,1)
Kb(3,3) = cd/DTIME+qYield*dzdu(2,2) + ke
ENDIF
ke=(G*Ar/Tr)*(1-(fb(1)/Fcrr)**2)

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C
get moment and stiffness in torsion about x, rotation about y, z
Kb(4,4)=Kt
Kb(5,5)=Kr
Kb(6,6)=Kr
fb(4)=Kb(4,4)*ub(4)
fb(5)=Kb(5,5)*ub(5)
fb(6)=Kb(6,6)*ub(6)
C
global stiffness matrix
Kg=MATMUL(MATMUL(MATMUL(TglT,TlbT),Kb),Tlb),Tgl)
C
global force matrix
Fg=MATMUL(TglT,MATMUL(TlbT,fb))
SVARS(25)=umax
SVARS(26:27)=z
SVARS(28)=ke
END SUBROUTINE
END

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